# UCI Rocket Project **Project Advisor: Professor Kenneth Mease**

#### Mission

Design and build a cost effective means of reaching space by using a Rockoon system with the ultimate goal of carrying a Cube Satellite to orbit.

#### Background

Traditional CubeSat launch costs to space for a sounding rocket cost \$40,000/Cube to \$2,000,000/Cube and launches to the mesosphere cost \$25,000/kg to \$40,000/kg. Currently there is a push from both Academia and Industry to develop these small satellites because of their low cost and proportional functionality compared to a large (full scale) satellite.

## Objectives

#### **Avionics:**

- 1. Precision landing System (04/16 3/16)
- 2. Avionics for Rocket (01/15-1/17)
- 3. Avionics for Rockoon (01/16-01/17)

#### **Structures:**

- 1. Finish Prototype 2 and Launch (10/15-1/16)
- 2. Rocket Airframe (11/16-5/17)
- 3. Large Rockoon, Prototype 3 (03/17-05/18) **Propulsion:**
- 1. Develop Hybrid Engine (01/16-2/17)
- 2. Successful Engine Test (2/17)
- 3. Mass Production and Flight Ready Engine (2/17-6/17)

Propulsion, \$2,995

## Budget

**Avionics**, \$2,237

Structures, \$4,879

#### Structures

**Team lead: Linh Ly** Members: Aleeza Roque, Norberto Abadias, Kevin Chen, Stephen Moes, Phong Huynh

#### **Avionics**

Team lead: Jesse Inoyue Members: Tarik Snyder, Joshua Yang, Zhiyang Feng, Roger Yao, Santiago Martin, Samuel Tse, Aroosa Ansari

Propulsion Team lead: Tai Wei Chen Members: Oziel Ortiz, Luis Gallegos, Omar Medina, Grant Wu, Justin Block

### Rocket

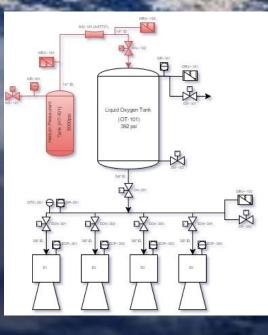
Sea Launch: ~45km; 147,000 ft Rockoon launch: ~270 km; 886,000ft Weight: ~140 kg (60kg of propellant)

## **Hybrid Chemical Rocket Engine**

- Thrust: 1000 lbf.
- Propellants: Liquid Oxygen (oxidizer) + High Density Polyethylene (fuel)
- Engine Made from Carbon Fiber Composite to drastically reduce weight

#### Four Engine Cluster





#### **Rockoon:**

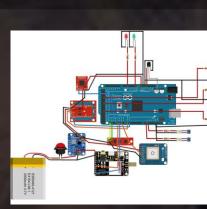
Tows rocket to 20km (65,000ft) before rocket launches. Replaces first stage of rocket in addition to giving the rocket an advantage in both initial height and operation in a less dense atmosphere

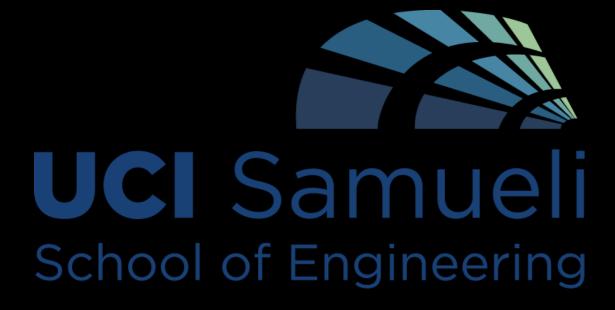
### **Prototype 1**

## **Prototype 2**

## Avionics

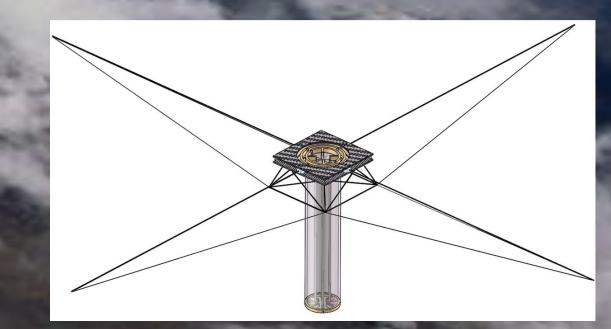
Capable of launching rocket, deploying parachute, and communicating with both Rocket and Rockoon. Must survive very cold environment and run on limited battery power to reduce weight. **Circuit Diagram** 





Cooled Ablatively using carbon ceramics similar to that used on the Space Shuttle





#### **Prototype 3**

**Precision Landing** System



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